編號:

146

國立成功大學一○○學年度碩士班招生考試試題

共2頁,第|頁

系所組別: 航空太空工程學系在職專班甲組

考試科目: 航空工程概論(專班)

考試日期:0219,節次:3

※ 考生請注意:本試題 ☑可 □不可 使用計算機

- At 12 km in the standard atmosphere, the pressure, density, and temperature are 1.9399x104 N/m², 3.1194x10-1 kg/m³, and 216.66 K, respectively. Assuming an isothermal region, calculate the standard atmospheric values of pressure, density, and temperature at an altitude of 18 km. (20%)
- 2. An airplane is flying at a standard altitude of 25,000 ft with density ratio to sea level density of 0.4481 and at a velocity of 700 ft/sec. How fast must the airplane be flying to experience the same dynamic pressure at sea level? (15%)
- A light airplane weighs 2,650 lbf and has a wing area of 174 ft². Estimate its sea level stall speed if the maximum lift coefficient is 1.8. (15%)

Note:

1.

R_{specific} for air Units

286.9 J kg⁻¹ K⁻¹

53.3533 ft lbf lb⁻¹ °R⁻¹

1716.59 ft lbf slug⁻¹ °R⁻¹

Based on a mean molar mass

for dry air of 28.9645 g/mol

2. The sea level standard value of air density: 1.229 kg/m^3 or $0..00237 \text{ slug/ft}^3$.

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- 4. (20%) Explain briefly the following terms and concepts:
 - (a) aerodynamic coefficients,
 - (b) Fatigue failure
 - (c) Composite materials
 - (d) Stress and strain
- 5. (10%) In aircraft design, what are the major concerns in selecting appropriate structural materials?
- 6. (20%) The figures below show an airplane flying in the vertical plane and its free body. The forces acting on the plane are its weight W, the thrust T exerted by its engines, and the aerodynamic forces, which are resolved into a component perpendicular to the path, the lift L, and a component parallel to the path, the drag D. The angle γ between the horizontal and the path is called the flight path angle, and α is the angle of attack. If $\gamma = 6^{\circ}$, D = 125 kN, L = 680 kN, and the mass of the airplane is 72,000 kg, what values of T and α are necessary to maintain steady flight?

